EFFECTS OF CANOPY GEOMETRY ON THE DRAG COEFFICIENT OF A CROSS PARACHUTE IN THE FULLY OPEN AND REEFED CONDITIONS FOR A W/L RATIO OF 0.264

Details of illustrations in this document may be become studied on microfiche

By W. P. Ludtke

20 AUGUST 1971



NAVAL ORDNANCE LABORATORY. WHITE OAK, SILVER SPRING, MARYLAND

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Prepared by: W. P. Ludtke

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NAVAL ORDNANCE LABORATORY WHITE OAK, MARYLAND

20 August 1971

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Effects of Canopy Geometry on the Drag Coefficient of a Cross Parachute in the Fully Open and Reefed Conditions for a W/L Ratio of 0.264

The investigation presented in this report is related to the improvement of parachute technology.

ROBERT ENNIS
-----Captain, USN
Commander

V. C. D. DAWSON By direction

CONTENTS

- The Control of th

																			Page
INTRODUCT	.IC	N			•	•		•						•		•		•	1
APPROACH					•		•					•							1
RESULTS.		•						•		•	•		•		•			•	3
APPENDIX	Α																•		A-1

ILLUSTRATIONS

or of the second substances of subside the second substances of the substance of the second subsiderations and the second subsiderations and the second subsiderations and the second subsiderations and the second subsiderations and the second subsiderations are second subsiderations.

Figure	
1	Model Parachute Configurations
2	Model Parachute Construction Details
3	Plan View of Wind Tunnel Support and Photographic Systems
4	Drag Coefficient Test Data; Parachute Series No. 2
5	Drag Coefficient Test Data; Parachute Series No. 3
6	Drag Coefficient Test; Parachute Series No. 2; 8 Suspension Line Parachute; Suspension Line Length 1.0 L
7	Drag Coefficient Test; Parachute Series No. 2; 8 Suspension Line Parachute; Suspension Line Length = 1.4 L
8	Drag Coefficient Test; Parachute Series No. 2; 8 Suspension Line Parachute; Suspension Line Length = 1.8 L
9	Drag Coefficient Test; Parachute Series No. 2; 16 Suspension Line Parachute; Suspension Line Length = 1.0 L
10	Drag Coefficient Test; Parachute Series No. 2; 16 Suspension Line Parachute; Suspension Line Length = 1.4 L
11	Drag Coefficient Test; Parachute Series No. 2; 16 Suspension Line Parachute; Suspension Line Length = 1.8 L
12	Drag Coefficient Test; Parachute Series No. 2; 24 Suspension Line Parachute; Suspension Line Length = 1.0 L
13	Drag Coefficient Test; Parachute Series No. 2; 24 Suspension Line Parachute; Suspension Line Length = 1.4 L

Figure	
14	Drag Coefficient Test; Parachute Series No. 2; 24 Suspension Line Parachute; Suspension Line Length = 1.8 L
15	Drag Coefficient Test; Parachute Series No. 3; 8 Suspension Line Parachute; Suspension Line Length = 1.0 L
16	Drag Coefficient Test; Parachute Series No. 3; 8 Suspension Line Parachute; Suspension Line Length = 1.4 L
17	Dræg Coefficient Test; Parachute Series No. 3; 8 Suspension Line Parachute; Suspension Line Length = 1.8 L
18	Drag Coefficient Test; Parachute Series No. 3; 16 Suspension Line Parachute; Suspension Line Length = 1.0 L
19	Drag Coefficient Test; Parachute Series No. 3; 16 Suspension Line Parachute; Suspension Line Length = 1.4 L
20	Drag Coefficient Test; Parachute Series No. 3; 16 Suspension Line Parachute; Suspension Line Length = 1.8 L
21	Drag Coefficient Test; Parachute Series No. 3; 24 Suspension Line Parachute; Suspension Line Length * 1.0 L
22	Drag Coefficient Test; Parachute Series No. 3; 24 Suspension Line Parachute; Suspension Line Length = 1.4 L
23	Drag Coefficient Test; Parachute Series No. 3; 24 Suspension Line Parachute; Suspension Line Length = 1.8 L
24	Reefed Canopy Test Data; Parachute Series No. 2, Test Velocity = 275 fps
25	Reefed Canopy Test Data; Parachute Series No. 3,

Figure	
26	Reefed Canopy Test; Parachute Series No. 2; 8 Suspension Line Parachute; Suspension Line Length = 1.0 L; Test Velocity = 275 fps
27	Reefed Canopy Test; Parachute Series No. 2; 8 Suspension Line Parachute; Suspension Line Length = 1.4 L, Test Velocity = 275 fps
28	Reefed Canopy Test; Parachute Series No. 2; 8 Suspension Line Parachute; Suspension Line Length = 1.8 L; Test Velocity = 275 fps
29	Reefed Canopy Test; Parachute Series No. 2; 16 Suspension Line Parachute; Suspension Line Length = 1.0 L; Test Velocity = 275 fps
30	Reefed Canopy Test; Parachute Series No. 2; 16 Suspension Line Parachute; Suspension Line Length = 1.4 L; Test Velocity = 275 fps
31	Reefed Canopy Test; Parachute Series No. 2; 16 Suspension Line Parachute; Suspension Line Length = 1.8 L; Test Velocity = 275 fps
32	Reefed Canopy Test; Parachute Series No. 2; 24 Suspension Line Parachute; Suspension Line Length = 1.0 L; Test Velocity = 275 fps
33	Reefe_ Canopy Test; Parachute Series No. 2; 24 Suspension Line Parachute; Suspension Line Length = 1.4 L; Test Velocity = 275 fps
34	Recfed Canopy Test; Parachute Series No. 2; 24 Suspension Line Parachute; Suspension Line Length = 1.8 L; Test Velocity = 275 fps
35	Reefed Canopy Test; Parachute Series No. 3; 8 Suspension Line Parachute; Suspension Line Length = 1.0 L; Test Velocity = 275 fps
36	Reefed Canopy Test; Parachute Series No. 3; 8 Suspension Line Parachute; Suspension Line Length = 1.4 L; Test Velocity = 275 fps
37	Reefed Canopy Test; Parachute Series No. 3; 8 Suspension Line Parachute; Suspension Line Length = 1.8 L; Test Velocity = 275 fps

Figure	
38	Reefed Canopy Test; Parachute Series No. 3; 16 Suspension Line Parachute; Suspension Line Length = 1.0 L; Test Velocity = 275 fps
39	Reefed Canopy Test; Parachute Series No. 3; 16 Suspension Line Parachute; Suspension Line Length = 1.4 L; Test Velocity = 275 fps
40	Reefed Canopy Test; Parachute Series No. 3; 16 Suspension Line Parachute; Suspension Line Length = 1.8 L; Test Velocity = 275 fps
41	Reefed Canopy Test; Parachute Series No. 3; 24 Suspension Line Parachute; Suspension Line Length 1.0 L; Test Velocity = 275 fps
42	Reefed Canopy Test; Parachute Series No. 3; 24 Suspension Line Parachute; Suspension Line Length = 1.4 L; Test Velocity = 275 fps
43	<pre>eefed Canopy Test; Parachute Series No. 3; 24 Suspension Line Parachute; Suspension Line Length = 1.8 L; Test Velocity = 275 fps</pre>
4 4	Drag Coefficient Test Data; Parachute Series No. 2; 8 Suspension Lines at 1.8 L Length

TABLES

Table	Title
I	Materials Used in Model Parachute Construction
II	Drag Coefficient Test Data; Parachute Series No. 2
III	Drag Coefficient Test Data; Parachute Series No. 3
Γ'	Reefed Canopy Test Data; Parachute Series No. 2, Test Velocity = 275 fps
v	Reefed Canopy Test Data; Parachute Series No. 3, Test Velocity = 275 fps

REFERENCE

1 AFFDL, AIRFORCESYSCOM, "Drag and Stability of Cross Type Parachutes," FDL-TDR-64-155, Feb 1965

SYMBOLS

drag force, 1bs D coefficient of drag c_{n} velocity, ft/sec density of air, slugs/ft3 dynamic pressure, 1bs/ft² canopy reference area, ft² length of canopy arm L width of canopy arm length of reefing line 1 canopy arm width-to-length ratio W/L reefing line length-to-canopy-diameter ratio

1/L

DEFINITIONS

Permeability

rate of airflow through cloth in $\rm ft^3/ft^2\text{-}min$ when measured under a pressure differential of 1/2 inch of water.

Skirt Reefing

a restriction of the skirt of a drag-producing surface to a diameter less than its diameter when it is fully inflated.

Percent Reefed ratio of the drag force produced in the reefed condition to the drag force of the fully inflated parachute.

INTRODUCTION

Limited supersonic wind-tunnel tests at Mach numbers of M=1.6, 1.8, 2, and 3.2 demonstrated that the cross-type parachute has positive inflation with predictable aerodynamic drag and very good stability characteristics. This, together with very good subsonic aerodynamic stability and drag efficiency, and a low infinite mass opening shock factor, indicates that the cross parachute can be a very useful high-performance decelerator. The basic simplicity of the design should allow for some reduction in cost compared to equivalent ribbon and ring slot configurations, provided similar manufacturing tolerances are applicable.

- 1. The problem areas which have been encountered with the cross parachute are:
 - a. Lack of good definition of the drag coefficient
 - b. Absence of data on the reefed canopy characteristics

Heinrich, in reference (a), investigated the effects of cloth effective porosity, arm width-to-length ratio (W/L), and angle of attack on the static stability and drag coefficients of the cross parachute. Experience with the cross parachute indicates that additional parameters, other than those considered in reference (a), affect the drag-producing capability of this design, namely, the number of suspension lines, suspension line length, and velocity.

- 2. The purposes of this investigation are:
- a. To determine the effects of geometric configuration on the drag coefficient of a Cross parachute having an arm width-to-length ratio of 0.264. The parameters investigated are cloth permeability, number of suspension lines; suspension line length, and velocity.
- b. To establish the percent reefing of the various parachute configurations as a function of reefing line length to canopy arm length ratio.

APPROACH. Three series of model cross parachutes were designed using a canopy cloth of different air permeability for each series. All models consisted of two panels 40 inches in length with a W/L = 0.264. The two panels were arranged to form the configurations

illustrated in Figure 1. Each series of models consisted of three parachutes with 8, 16, and 24 suspension lines, respectively, for the same canopy cloth. As initially installed, the suspension lines were 1.8 canopy diameters in length. These lines were later shortened to 1.6 and 1.4 canopy diameters. This approach provided 27 possible geometric configurations for drag coefficient studies. Installation of reefing rings in the skirt hem provided an additional 27 reefed parachute configurations. Parachute construction details are illustrated in Figure 2 and the materials used in construction of the models are enumerated in Table I.

3. The wind-tunnel tests were conducted at the University of Maryland 7-foot x 11-foot cross section Subsonic Wind Tunnel at College Park, Maryland. The wind-tunnel support system, Figure 3, was designed to position the model canopies. A guide tube along the wind-tunnel center line permitted the control of parachute oscillations. To maintain a relatively aerodynamically uncluttered test section, guy wires were used to support the guide tube. In all tests, the parachute suspension lines were attached to the support ring of an aerodynamic drag force sensing device. Assembly in this manner lengthened the suspension lines of the various canopies to the required length. Each parachute was mounted on the support system, and measurements of the drag force were made at various wind-tunnel velocities from 50 fps through 300 fps. Reefing lines of 1/16-inch diameter flexible steel cable were then installed, and measurements of the drag force in the reefed configuration were made for several reefing line length-to-canopy-diameter ratios from 0.45 through 1.6. Upon completion of these tests, the parachute suspension lines were shortened to the next test length and the measurement procedures repeated.

4. Test data were reduced to coefficient form by means of the following formulae:

$$C_{D} = \frac{D}{qSo}$$

$$q = 1/2 \rho V^{2}$$

$$So = 2LW - W^{2}$$

- % reefed = Drag of parachute in reefed condition at velocity V

 Drag of fully opened parachute at same velocity
- 5. The reference area of all parachute models used in this test is 5.092 ft².

RESULTS

The experiments documented in this report have established the drag coefficients and reefed parachute characteristics for the cross parachute (W/L = 0.264) for various combinations of velocity, cloth permeability, suspension line length, number of suspension lines, and hem reefing line length. Of the three series of parachutes which were tested, meaningful data were obtained only on the number 2 and number 3 series canopies. Data from the series number 1 parachutes (cloth permeability of 8 ft3/ft2 min)were very limited due to the induced canopy rotation which resulted in the canopy spinning closed around the guide tube support system. Low cloth permeability appears to be another cause of canopy rotational instability. Series number 2 and number 3 (cloth permeability of 80 $\rm ft^3/ft^2$ min and 208 $\rm ft^3/ft^2$ min, respectively) remained fully open throughout the velocity test range. Drag coefficient data for the various fully opened configurations of the number 2 and number 3 series parachutes are tabulated in Tables II and III and graphically presented in Figures 4 and 5, respectively. These data show that for any given configuration, the lower permeability series number 2 parachutes have a higher drag coefficient than the series number 3. In all configurations, an increase in the suspension line length or the number of suspension lines was accompanied by an increase in drag coefficient. The drag coefficients of the eight suspension line canopies are essentially constant over the velocity range An increase in the number of suspension lines not only raises the magnitude of the drag coefficient, but also produces a drag rise with increasing velocity. There is a strong indication that the drag coefficient rises sharply at velocities less than 40 fps. An example of this effect is shown in Figure 44, Appendix A. The range of drag coefficients for the tested configurations varied from a minimum of 0.54 to a maximum of 0.75. Photographs of the fully inflated parachutes at wind-tunnel velocities of 50, 100, and 200 fps are presented in Figures 6 through 23. All parachutes were reefed using a 1/16-inch diameter flexible steel cable. Data were obtained for ratios of reefing line length-to-canopy-diameter of 0.45, 0.7, 0.85, 1.0, 1.15, and 1.6. These data are tabulated in Tables IV and V and graphically represented in Figures 24 and 25. Since the drag of the fully inflated parachute increases as the number and/or length of suspension lines increases, the percent reefed for a given reefing line lengthto-canopy-diameter ratio is reduced.

TABLE I

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MATERIALS USED IN MODEL PARACHUTE CONSTRUCTION

ITEM	MATERIAL		PARACHUTE SERIES	
		NUMBER 1	NUMBER 2	NUMBER 3
-	СІОТН	MIL-C-7020, TYPE 1 HEAT SET, AIR PERMEABILITY 8FT ³ /FT ² /MIN * 1/2 INCH WATER FRESSURE DIFFERENTIAL	MIL-C-7020, TYPE I	4.75 OZ./YD ² , DOBBY WEAVE, AIR PERMEABILITY 208 FT ³ /FT ² /MIN = 1/2 INCH WATER PRESSURE DIFFERENTIAL
2	TAPE	MIL-1-5038, TAPE III, 1/2 INCH WIDE	MIL-T-5038, TYPE III, 1/2 INCH WIDE	MIL-T-5038, TYPE III, 1/2 INCH WIDE
3	SUSPENSION ¹ LINE	MIL-C-17183	MIL-C-17183	MIL-C-17183
4	STITCHES ²	TYPE 301, FED STS 751, 9 TO 12 STITCHES PER INCH, 2 ROWS ON 1/4 INCH NEEDLE GAUGE.	CHES PER INCH, 2 ROWS	ON 1/4 INCH NEEDLE GAUGE.
5	STITCHES	TYPE 301, FED STD 751, 9 TO 12 STITCHES PER INCH, SINGLE ROW	CHES PER INCH, SINGLE R	NOW

8 SUSPENSION LINE CANOPIES USE, TYPE VI, 500 LB TENSILE STRENGTH 16 SUSPENSION LINE CANOPIES USE, TYPE IV, 300 LB TENSILE STRENGTH 24 SUSPENSION LINE CANOPIES USE, TYPE III, 200 LB TENSILE STRENGTH

ALL THREAD, V-T-295. TYPE I OR II, CLASS I OR 2, SIZE B

TABLE II

DRAG COEFFICIENT TEST DATA; PARACHUTE SERIES NO. 2

		Z LINES	E LENGTH	METER	1.8 L	0.670	089.0	0.704	0.721	0.738	0.749	0.757	0.762	0.775	0.774	0.779
		24 SUSPENSION LINES	SUSPENSION LINE LENGTH	CANOPY DIAMETER	1.4 L	0.645	0.645	0.661	0.682	0.700	0.704	0.714	0.715	0,726	0.729	0.729
		24 51	SUSPEN	S	1.0 [0.595	0.587	0.598	0.603	0.613	0.626	0.638	0.641	0.651	0.658	0.662
	EN T	LINES	ENGTH	ETER	1.8 L	0.620	0.646	0.677	189.0	0.694	0.701	0.715	0.724	0.730	0.732	0.733
	DRAG COEFFICIENT	16 SUSPENSION LINES	SUSPENSION LINE LENGTH	CANOPY DIAMETER	1.4 L	0.620	0.629	0.624	0.644	0.667	0.672	0.679	0.685	0.692	0.695	869.0
,	DRA(NS 91	SUSPENS	S S	1.0.1	195.0	9.577	0.580	0.589	0.605	0.603	0.613	6.617	919.0	0.624	0.625
		NES	ENGTH	TER	1.8 ℃	0.604	9.610	0.635	0.638	0.641	0.637	0.645	0.642	0.650	0.645	0.654
		8 SUSPENSION LINES	Suspension line length	CANOPY DIAMETER	1.4 L	0.632	0.611	0.625	0.627	0.627	0.629	0.633	0.627	0.631	0.633	0.615
		8 SUS	SUSPENS	CAN	1.0 L	0.540	0.566	0.568	0.560	0.566	0.568	0.570	0.566	0.568	0.570	0.568
			DYNAM!C PRESSURE	o -		2.973	6.689	11.892	18.582	26.758	36.420	47.569	60.602	74.327	89.936	107.031
			VELOCITY	>		50	75	92	125	150	175	200	225	250	275	300

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TABLE III DRAG COEFFICIENT TEST DATA, PARACHUTE SERIES NO. 3

		24 SUSPENSION LINES	SUSPENSION LINE LENGTH	CANOPY DIAMETER	1.4.[+			0 700	, , ,	0.724	0.734	0.734	0.738	0.741	0.745
		24 SUSPENSION	SUSPENSION LINE	ANOPY DIAME		789	83									
		24 SUS	SUSPENS	Z		9	0.683	0 669	0.685	0.679	689	0.701	0.702	0.710	0.715	0.722
,		i —		٥	1.0 [0.624	0.599	0.599	0.608	0.612	0.616	0.618	0.625	0.632	0.639	0.645
	ENI	L'NES	ENGTH	TER	1.8.1	0.6%	0.677	0.686	0.692	0.693	0.700	0.709	0.706	0.712	0.717	0.722
	DRAG COEFFICIENT	16 SUSPENSION L'NES	SUSPENSION LINE LENGTH	CANOPY DIAMETER	1.4 [0.645	0.650	0.645	0.652	0.662	0.659	0.665	899.0	0.671	0.677	0.679
400	DRA	NS 91	SUSPEN	S	1.0.1	0.570	0.594	0.583	0.586	0.585	0.580	0.589	0.593	0.605	0.617	0.623
		ZES	LENGIH	ETER	1.8 L	0.708	0.654	0.650	0.00	0.661	0.659	0.655	0.656	0.656	0.657	0.659
		8 SUSPENSION LINES	SUSPENSION LINE LENGTH	CANOPY DIAMETER	1.4 [0	0.670	0.646	0.636	0.645	0.635	0.634	0.630	0.634	0.634	0.637
		8 SUSP	SUSPEN	Y Z	1.0 [0.599	0.598	0.588	0.588	0.587	0.586	0.585	0.584	0.583	0.584	0.584
		PRESSURE	0		!	2.973	6.689	11.892	18.582	26.758	36.420	47.569	60.602	74.327	89.936	107.031
		VELOCITY	>			20	75	8	125	150	175	200	225	256	275	300

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TABLE IV
REEFED CANOPY TEST DATA; PARACHUTE SERIES NO. 2
TEST VELOCITY - 275 FPS

				PE	PERCENT REFFED	ED			
REFING LINE LENGTH	8 SU	8 SUSPENSION LINES	INES	NS 91	16 SUSPENSION LINES	LINES	24 31	24 SUSPENSION LINES	LINES
CANCPY DIAMETER	SUSPEN	SUSPENSION LINE LENGTH	LENGTH	SUSPEN	SUSPENSION LINE LENGTH	LENGIH	SUSPEN	SUSPENSION LINE LENGTH	LENGIH
~ -	CA	CANOPY DIAMETER	ETER	CA	CANOPY DIAMETER	ETER	CA	CANOPY DIAMETER	Erer
,	1.0 L	1.4 L	1.8.1	1.0 L	1.4 [1.8 L	1.0.1	1.4 L	181
1.60	98.1	4.96	94.0	8.96	9.88	84.2	93.8	85.1	80.9
1.30	6.28	78.8	76.3	81.1	74.2	1.69	75.7	0.69	65.5
1.15	80.5	73.1	71.0	72.3	65.5	62.6	67.9	61.6	58.7
0.1	72.5	9.99	64.8	64.7	58.3	55.6	61.1	55.0	52.3
0.85	8.18	54.0	52.7	56.3	50.4	48.2	54.1	48.3	46.0
0.70	46.1	40.6	40.0	44.6	40.9	38.5	44.8	39.3	37.1
0.45	31.5	27.6	26.4	26.6	23.7	24.0	26.3	22.9	21.4

TABLE V

The second secon

REEFED CANOPY TEST DATA; PARACHUTE SERIES NO. 3 TEST VELOCITY = 275 PPS

	24 SUSPENSION LINES	HIS AND MAN TO SELECT	SUSPENSION LINE LENGTH	CANCAT ULANELLE	1.01 1.41 1.81		83.4		70.1 63.0 61.1	62.3 55.8 54.4	53.5 48.0 46.8	43.8 39.2 38.7	23.6 21.3 21.5		
PERCENT REFFED		16 SUSPENSION LINES	SUSPENSION LINE LENGTH	CANOPY DIAMETER		1.01 1.46				73.9 66.6 63.7				22.7 21.0 20.1	
		SINI NOISINESS	8 SUSPENSION EN EN	SUSPENSION LINE LENGTH	CANOPY DIAMETER	141 1.81	70'	07 8 94.7	78.9	7.17	61.7		37.1 36.8		
		, , , , , , , , , , , , , , , , , , ,	HENGT TINE LENGTH	KEELING STORY DIAMETER	CANCT	~ -			1.60	1.30	1.15	1.00	0.85	0.70	0.45

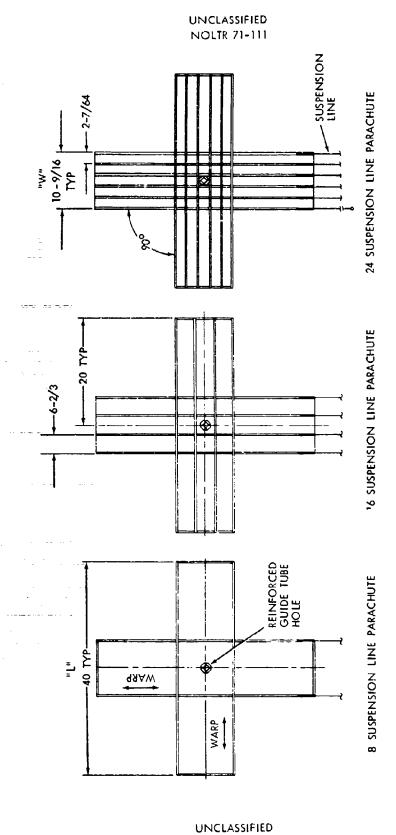
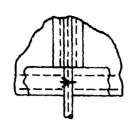


FIG. 1 MODEL PARACHUTE CONFIGURATIONS-CONSTRUCTION DETAILS ARE SHOWN IN FIG. 2

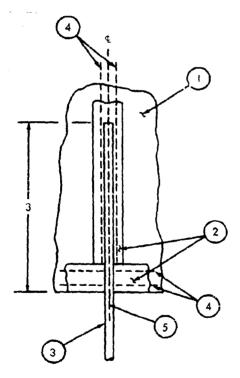


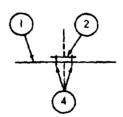


REEFING RING TIE CORD



1/4 O.D. x 3/16 I.D. x 1/8 LONG

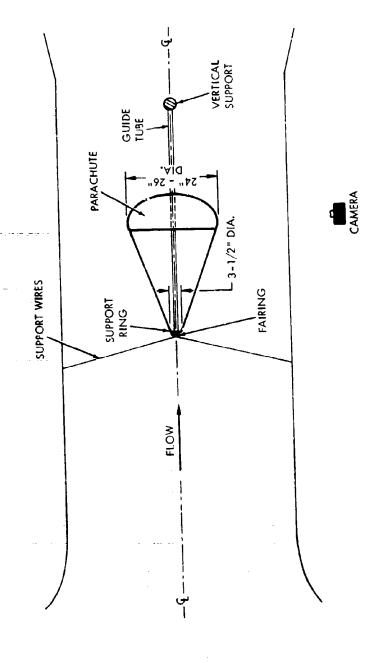




TYPICAL TAPE - CANOPY CROSS SECTION

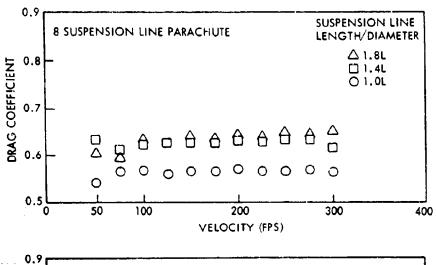
SKIRT HEM - SUSPENSION LINE ASS'Y

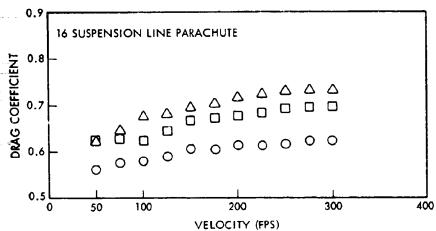
FIG. 2 MODEL PARACHUTE CONSTRUCTION DETAILS SEE TABLE I FOR MATERIALS IDENTIFICATION



WIND TUNNEL CROSS SECTION DIMENSIONS 7 FT x 11 FT FIG. 3 PLAN VIEW OF WIND TUNNEL SUPPORT AND PHOTOGRAPHIC SYSTEMS







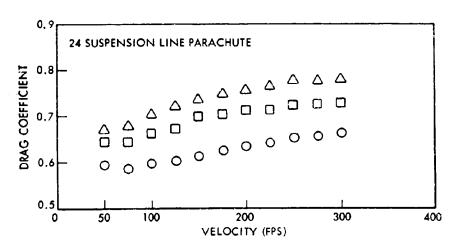
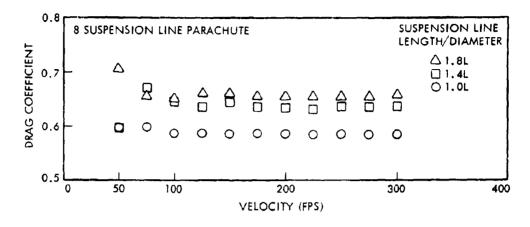
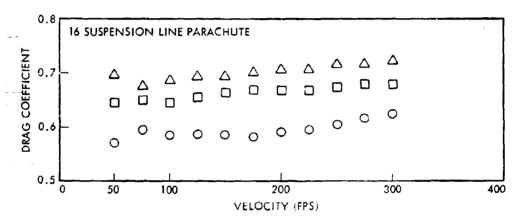


FIG. 4 DRAG COEFFICIENT TEST DATA; PARACHUTE SERIES NO. 2





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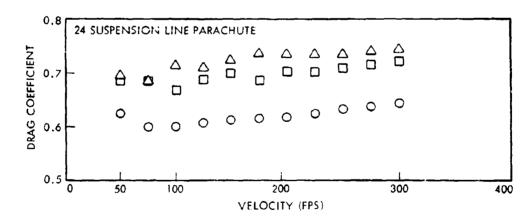


FIG. 5 DRAG COEFFICIENT TEST DATA; PARACHUTE SERIES NO. 3

VELOCITY = 50 FPS



VELOCITY = 100 FPS

VELOCITY = 200 FPS

FIG. 6 DRAG COEFFICIENT TEST; PARACHUTE SERIES NO. 2; 8 SUSPENSION LINE PARACHUTE; SUSPENSION LINE LENGTH = 1.0 L

VELOCITY = 50 FPS

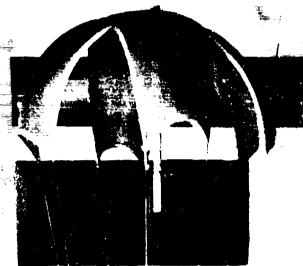
VELOCITY = 100 FPS

VELOCITY = 200 FPS

FIG. 7 DRAG COEFFICIENT 1EST; PARACHUTE SERIES NO. 2; 8 SUSPENSION LINE PARACHUTE; SUSPENSION LINE LENGTH - 1.4 L



VELOCITY = 50 FPS



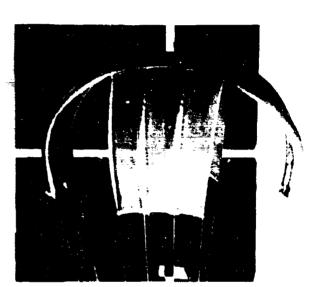
VELOCITY = 100 FPS



VELOCITY = 200 FPS

FIG. 8 DRAG COEFFICIENT TEST; PARACHUTE SERIES NO. 2; 8 SUSPENSION LINE PARACHUTE; SUSPENSION LINE 1 FINGTH = 1.8 L





VELOCITY = 100 FPS

VELOCITY = 200 FPS

FIG. 9 DRAG COEFFICIENT TEST; PARACHUTE SERIES NO. 2; 16 SUSPENSION LINE PARACHUTE; SUSPENSION LINE LENGTH = 1.0 L

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VELOCITY = 100 FPS

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VELOCITY = 200 FPS

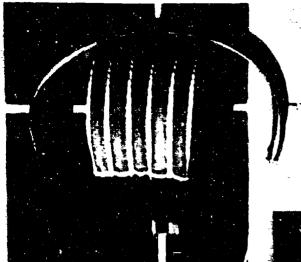
FIG. 10 DRAG COEFFICIENT TEST; PARACHUTE SERIES NO. 2; 16 SUSPENSION LINE PARACHUTE; SUSPENSION LINE LENGTH = 1.4 L

YELOCITY = 50 FPS

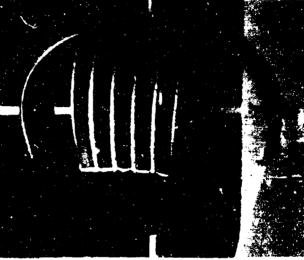
VELOCITY = 100 FPS

VELOCITY = 200 FPS

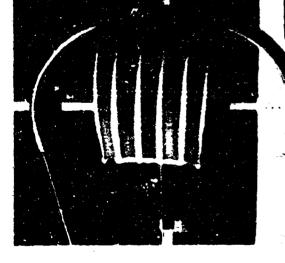
FIG. 11 DRAG COEFFICIENT TEST; PARACHUTE SERIES NO. 2; 16 SUSPENSION LINE PARACHUTE SUSPENSION LINE LENGTH 11.8 L



VELOCITY = 50 FPS

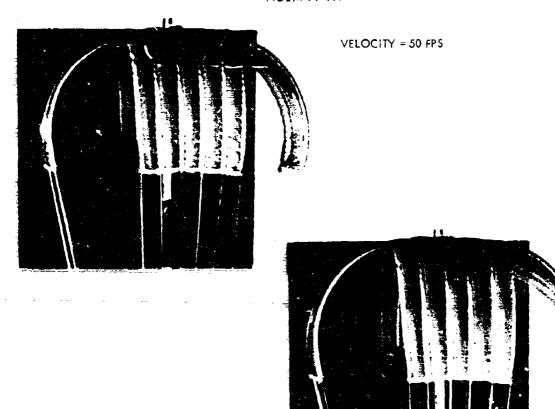


VELOCITY = 100 FPS



VELOCITY = 200 FPS

FIG. 12 DRAG COEFFICIENT TEST PARACHUTE SERIES NO. 2; 24 SUSPENSION LINE PARACHUTE; SUSPENSION LINE LENGTH = 1.0 L





VELOCITY = 200 FPS

FIG. 13 DRAG COEFFICIENT TEST; PARACHUTE SERIES NO. 2; 24 SUSPENSION LINE PARACHUTE; SUSPENSION LINE LENGTH = 1.4 L

VELOCITY = 50 FPS

VELOCITY = 100 FPS

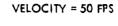
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VELOCITY = 200 FPS

FIG. 14 DRAG COEFFICIENT TEST; PARACHUTE SERIES NO. 2; 24 SUSPENSION LINE PARACHUTE; SUSPENSION LINE LENGTH= 1.8 L







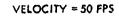
VELOCITY = 100 FPS

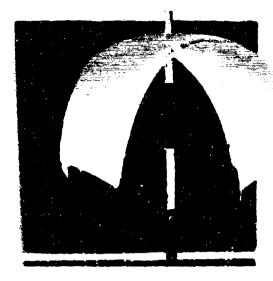


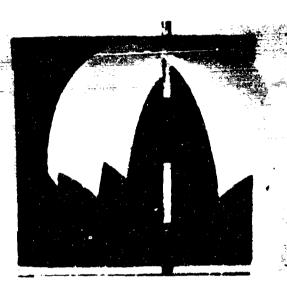
VELOCITY = 200 FPS

FIG. 15 DRAG COEFFICIENT TEST; PARACHUTE SERIES NO. 3; 8 SUSPENSION LINE PARACHUTE; SUSPENSION LINE LENGTH = 1.0 L

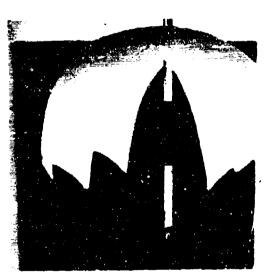
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VELOCITY = 100 FPS



VELOCITY = 200 FPS

FIG. 16 DRAG COEFFICIENT TEST; PARACHUTE SERIES NO. 3; 8 SUSPENSION LINE PARACHUTE; SUSPENSION LINE LENGTH = 1.4 L

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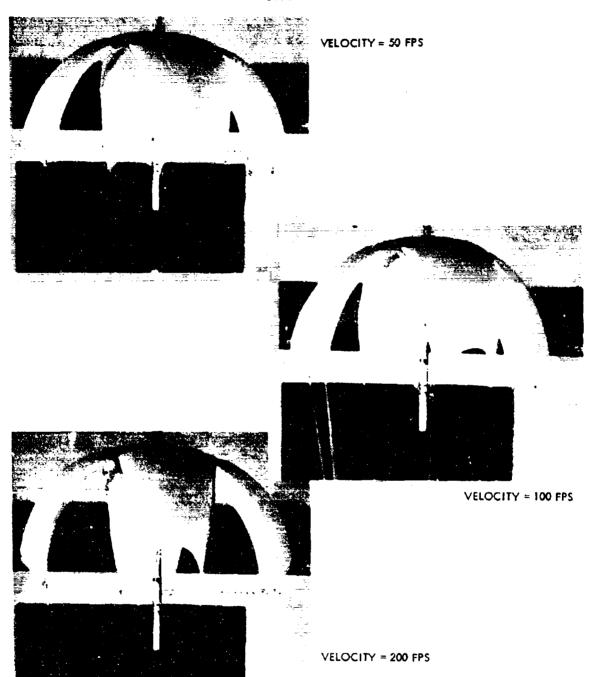
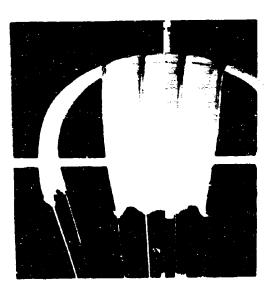


FIG. 17 DRAG COEFFICIENT TEST; PARACHUTE SERIES NO. 3; 8 SUSPENSION LINE PARACHUTE; SUSPENSION LINE LENGTH = 1.8 L



VELOCITY = 50 FPS





VELOCITY = 100 FPS

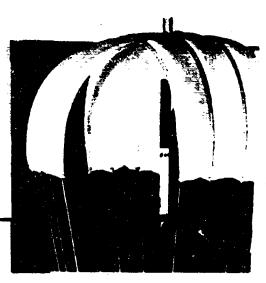
VELOCITY = 200 FPS

FIG. 18 DRAG COEFFICIENT TEST; PARACHUTE SERIES NO. 3; 16 SUSPENSION LINE PARACHUTE; SUSPENSION LINE LENGTH = 1.0 L

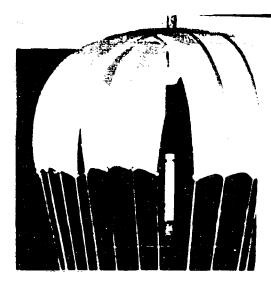
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VELOCITY = 50 FPS



VELOCITY = 130 FPS

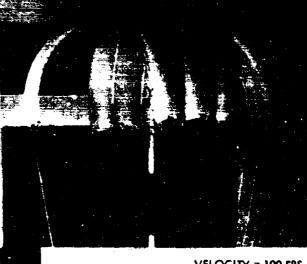


VELOCITY = 200 FPS

FIG. 19 DRAG COEFFICIENT TEST; PARACHUTE SERIES NO. 3; 16 SUSPENSION LINE PARACHUTE; SUSPENSION LINE LENGTH = 1.4 L



VELOCITY = 50 FPS



VELOCITY = 100 FPS

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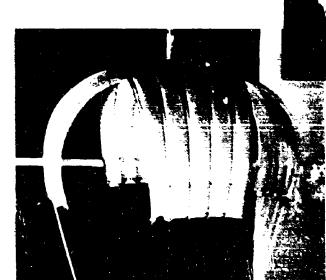
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VELOCITY = 200 FPS

DRAG COEFFICIENT TEST; PARACHUTE SERIES NO. 3; 16 SUSPENSION LINE PARACHUTE; SUSPENSION LINE LENGTH = 1.8 L $\,$ FIG. 20

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VELOCITY = 50 FPS

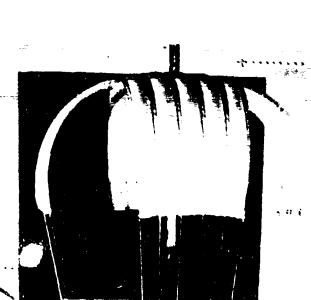


VELOCITY = 100 FPS

VELOCITY = 200 FPS

FIG. 21 DRAG COEFFICIENT YEST, PARACHUTE SERIES NO. 3; 24 SUSPENSION LINE PARACHUTE; SUSPENSION LINE LENGTH = 1.0 L





VELOCITY = 100 FPS

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VELOCITY = 200 FPS

FIG. 22 DRAG COEFFICIENT TEST; PARACHUTE SERIES NO. 3; 24 SUSPENSION LINE PARACHUTE; SUSPENSION LINE LENGTH = 1.4 L

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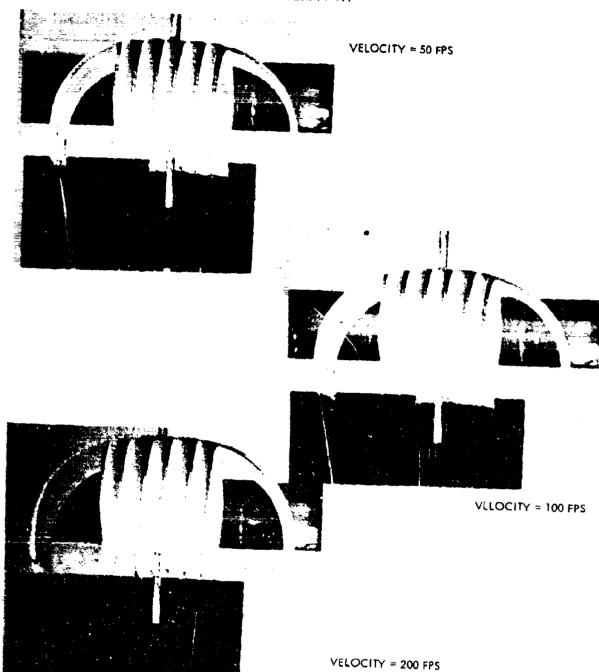
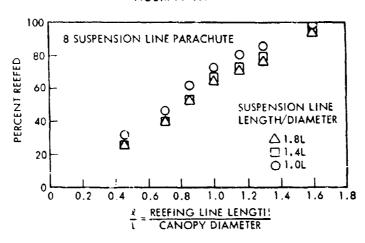


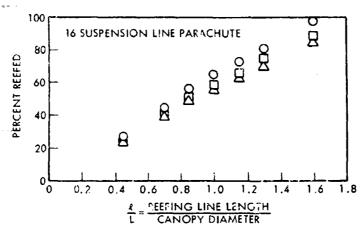
FIG. 23 DRAG COEFFICIENT TEST; PARACHUTE SERIES NO. 3; 24 SUSPENSION LINE PARACHUTE: SUSPENSION LINE LENGTH = 1.8 L



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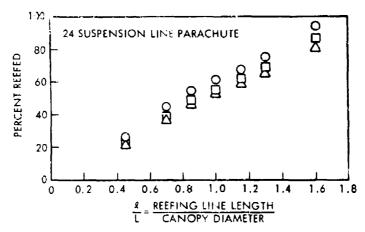
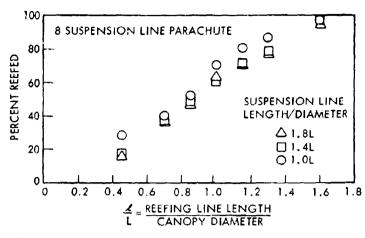
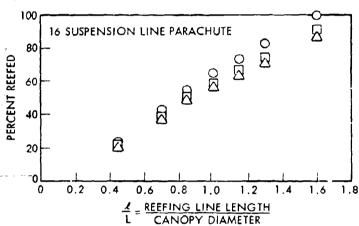


FIG. 24 REEFED CANOPY TEST DATA; PARACHUTE SERIES NO. 2
TEST VELOCITY = 275 FPS







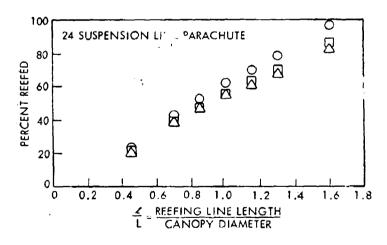


FIG. 25 REEFED CANOPY TEST DATA; PARACHUTE SERIES NO. 3
TEST VELOCITY = 275 FPS

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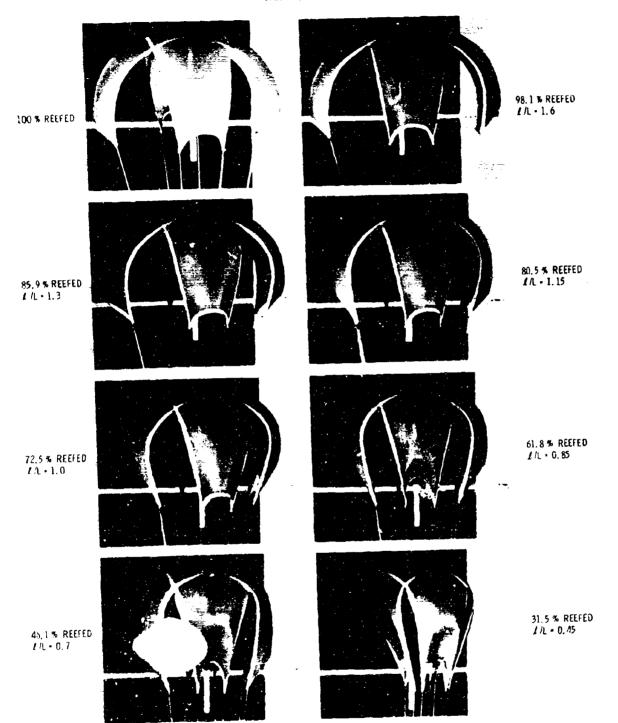


FIG. 15 SHIP TO A MICE OF TEXT PARACHUM, SHIPS TWO, 25 X SHISPENSION LINE PARACHUM, SHIPS SHOWN LINE LODGET 1, 311- TEXT VEGOCITY - 275 FPS





96.4 % REETED 1/L - 1.6

78.8 % REEFED //L • 1.3





73.1 % REEFED 1/L = 1.15







54.0% REEFED 1/L = 0.85







27, 6 % REE: ED 7/L • 0,45

FIG. 27 REFFED CANOPY TEST: PARACHUTE SERIES (I.O. 2- & SUSPENSION LINE PARACHUTE-SUSPENSION LINE LEGGED - 3, 31 TEST VELCOITY - 275 LPS



100 % REEFEC



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//L • 1.15





52.7 % REEFED i/L + 0.85





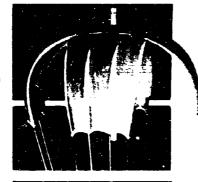


24,4% REEFED i/L • 0.45





THE 2- PROTECTION OF SERVICE THE SERVES WELL AS SUSPENSION LINE PROCEDULE SISSENSION LINE FROM THE NEED VELOCITY + 275 FPS





95,8 % REEFED

81,1 % REEFED 1/L = 1.3

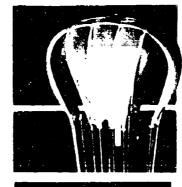




72.3 % REEFED 1/L = 1.15

64.7 % REEFEO





56.3 % REEFED 1/L = 0.85

44.6% REEFED //L = 0.7





26, 6 % REEFED 1/L + 0, 45

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EVEL 559 R.O.





88.6 % REEFED 1/L - 1.6



100 % REEFED





65.5 % REEFED 1/L - 1.15

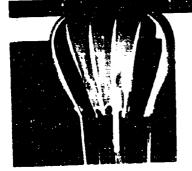


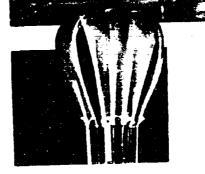




50.4% REEFED 1/L = 0.85



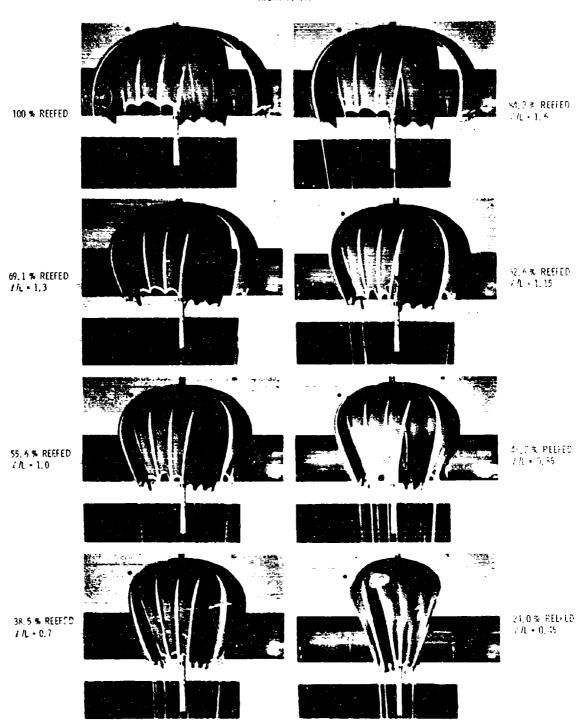




23.7 % REEFED

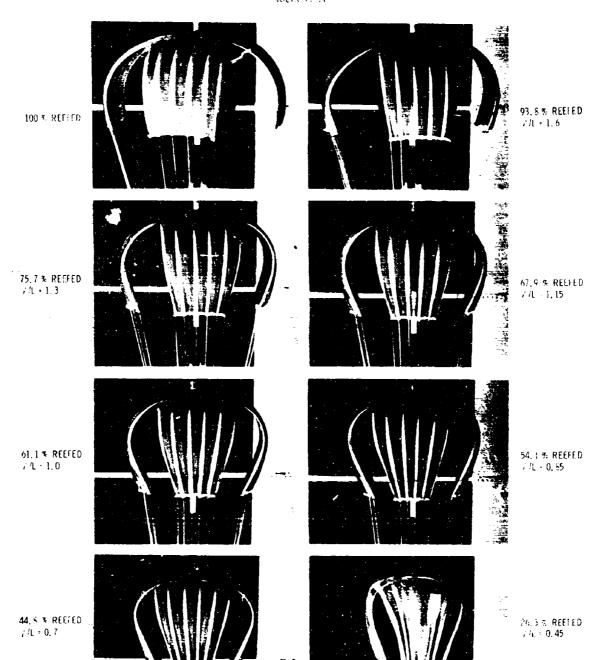
FIG. 30 RESERVED A TEST PARACHOLIC SERVES TO THE FESS SPEASION OF A SERVES ENGLISHED SERVES TO THE TEST VEGOCITY TO SEPS

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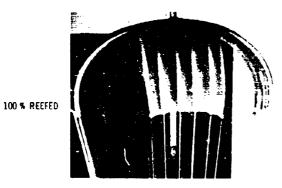
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where Σ , we can use the first transfer of the contribution of

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\$5.1 **% REEFED** ./(+ 1.6

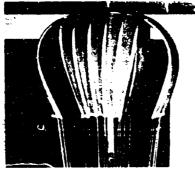






61.6% REEFED 7/L + 1.15

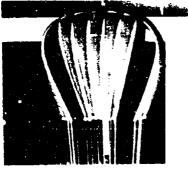






4×.3 % REEFED







22,9 % REEFED 1/0, ± 0,45

FIG. 35 REFER CAMOPY TESTS PARTCHOTE SERVICE CO. 24 SUSPENSE A CHARACTER SERVICE REPORT OF THE TEST AND A SERVICE AND A PO

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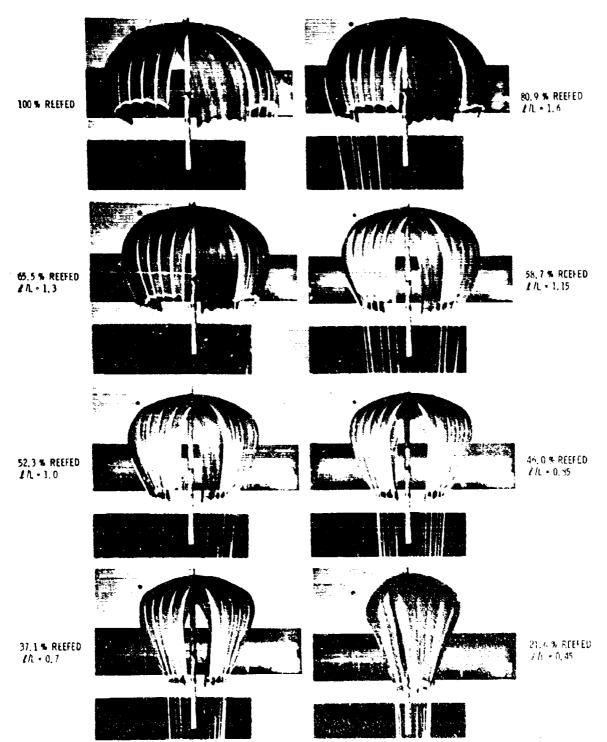


FIG. 34 RELECT CONORS TEST, PARACHER SCRIES NO. 2-71 SECRETSHOS HAVE PARACHER SUSPINISHED LITTLE HOLDER 1.31 TEST VEHICLE 1.25 FPS





97,8 % REEFED 1/L + 1,6





81, 1 % REEFED 1/L = 1,15

70,7 % REEFED 1/L = 1,00

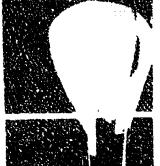




52.7 ≪ REEFED 7/L + 9.85

40.5 % REEFED 1/L + 0.7





28.1 % RODEB 7/L > 6.49

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97.8 % REEFED # // • 1.6



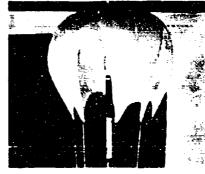




71.1 % REEFED 1 /L - 1.15

61.7 % REEFED 2/L - 1.0

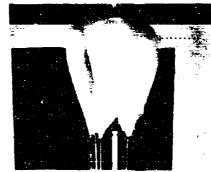




48, 7 % REEFED 1/L = 0, 85

37.1 % REEFED 2 1 - 0.7





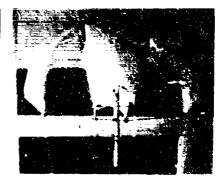
16.2 % REEFED 1/L - 0.45

Harden Commence of the Commenc

FIG. 36 RELEED CANOPY TEST: PARACHUTE SERIES NO. 3: 8 SUSPENSION LINE PARACHUTE: SUSPENSION LINE LENGTH - 1, 41: TEST VELOCITY - 275 FPS



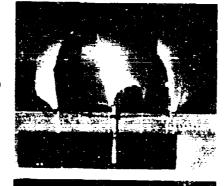






77.4 % REEFED 1/L = 1.3

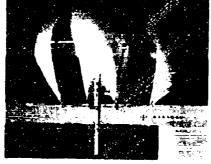
(金数)が無く、(数)に、(注意)記録しては就論した。 (左蓋)にして管理数(ました基準にした)だという。





71.1 % REEFED Z/L = 1,15

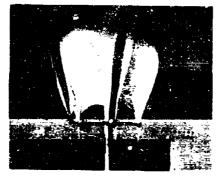
63,8 % REEFED //L - 1.0





47.4 % REEFED /L = 0, 85

36.8 % REEFED 1/L - 0.7



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:/L • 0.45

FIG. 37 REFED COMMENTEST PARTON TEST STRAIN TO THE SENSETTS FOR THE PARACHUTE SUSPENSION OF FEBRUARY 15 FOR THE STRAIN TO THE SENSET STRAIN TO THE SENSET STRAIN THE SENSET ST

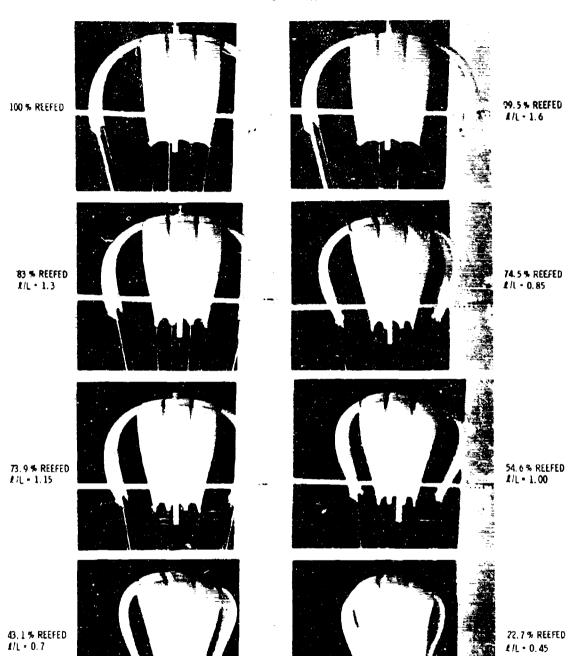


FIG. 3: REEFED CANOPY TEST, PARACHUTE STRIFTS NO. 3: 16 SUSPENSION LINE PARACHUTE, SUSPEN JON LINE LENGTH. 1.001; TEST VELOCITY + 275 FPS

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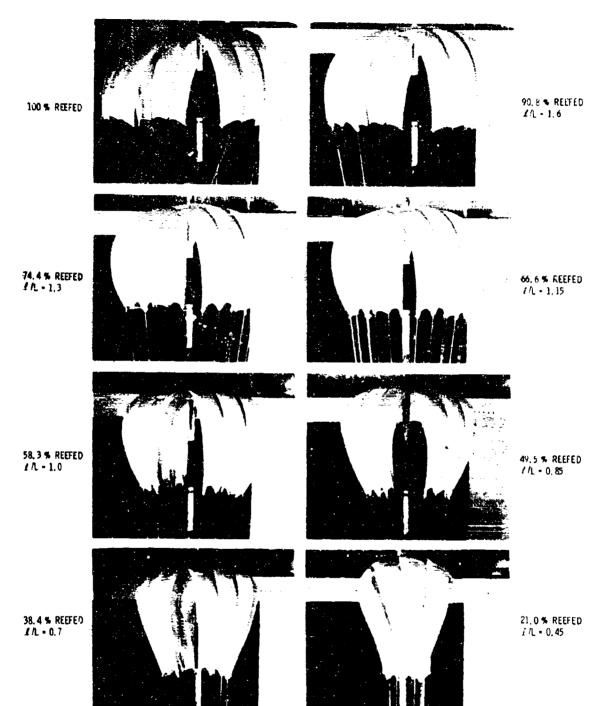
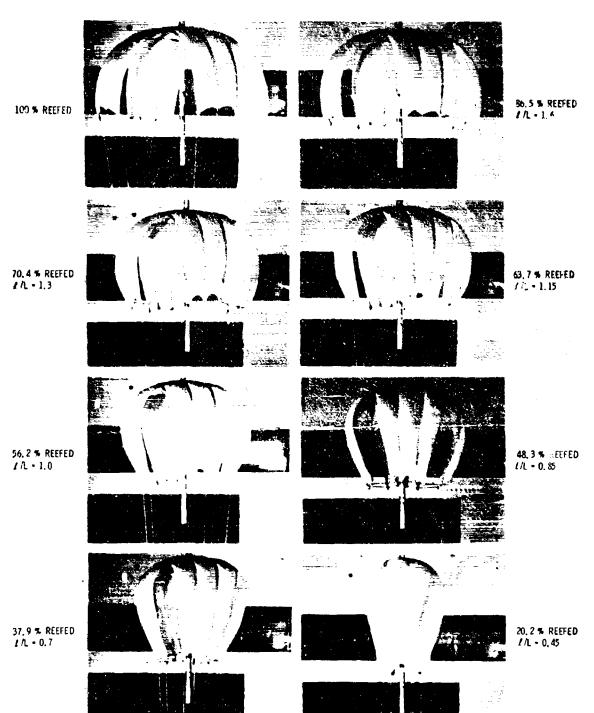


FIG. 39 REEFED CANOPY TEST: PARACHUTE SERIES NO. 19 10 SUSPENSION LINE PARACHUTE: SUSPENSION LINE LENGTH + 1,4 to TEST VELOCITY + 275 FPS



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FIG. 40 REEFED CANOPY TEST: PARACHUTE SERIES NO. 3; 16 SUSPENSION LINE PARACHUTE: SUSPENSION LINE LENGTH + 1.8 L: TEST VELOCITY + 275 FPS

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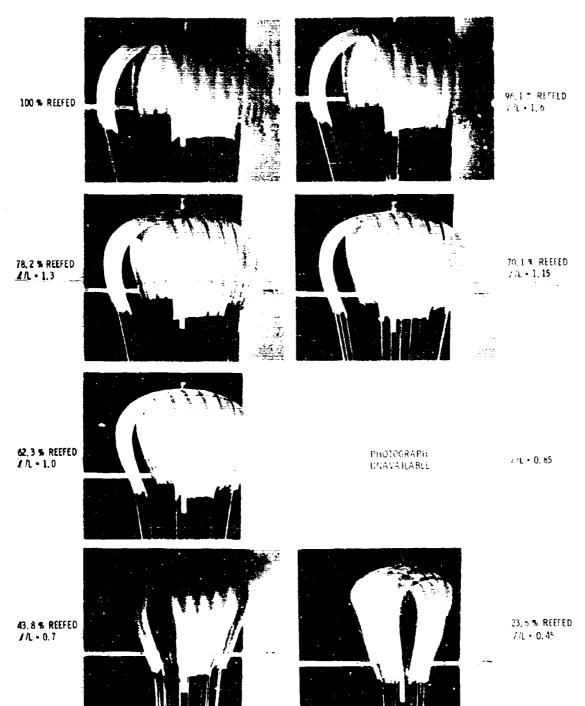
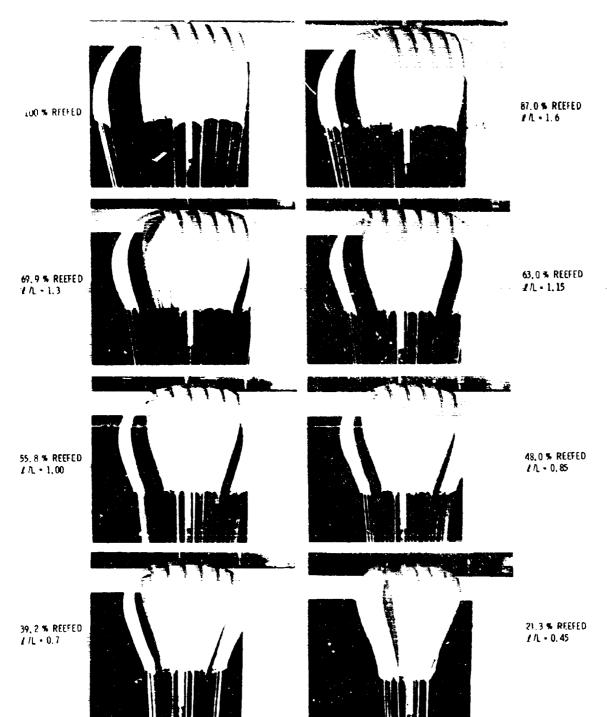


FIG. 41 RELIED CANOPY TUST: PARACHUTE STRIES NO. 3-24 SUSPENSION LINE PARACHUTE: SUSPENSION LINE (1907) - 1 PL. TEST VENCITY - 275 FPS

ा केंद्रिश्चाक्षक का क्षामीक्षरीय कर कर महीतिक्षित किया महत्त्र कर जीकिया कर है। इसिन्धिक कर होता है के

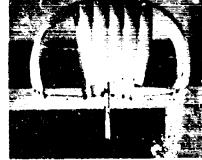
भागका स्टाहर का हास्त्रीत करात्रीहरू कर स्टब्स्स के महिल्ला का महिल्ला का महिल्ला का स्टाहरू के प्राप्त कर स्ट



THE 42 REPED CAMOPY DEST- PARACHUTE SERIES NO. 3-24 SUSPENSION LINE PARACHUTE; SUSPENSION LINE (EDIGTH+1, 4)- TEST VELOCITY + 275 FPS



100 % REEFED



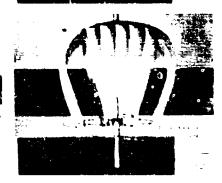
83.4 % REEFED 4/L - 1.6





61.1 % REEFED





46, 8 % REEFED 1/L + 0, 85







21.5 % REEFED 1/L = 0,45

FIG. 43 REEFED CANOPY TEST: PARACHUTE SERIES NO. 3: 24 SUSPENSION LINE PARACHUTE; SUSPENSION LINE (ENOTH-1.81-TEST VELOCITY + 275 FPS

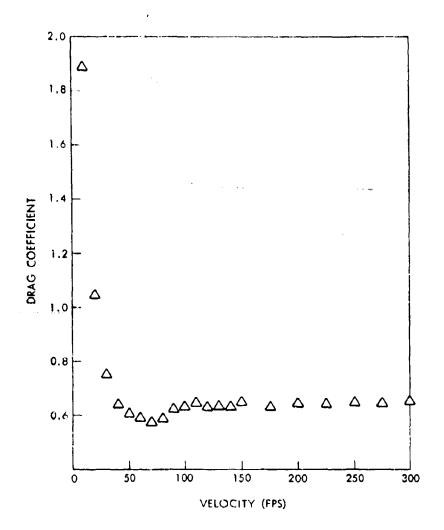


FIG. 44 DRAG COEFFICIENT TEST DATA; PARACHUTE SERIES NO. 2, 8 SUSPENSION LINES AT 1.8 L LENGTH

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APPENDIX A

Initially, the investigations of the parachute drag characteristics were to encompass a velocity range from 10 fps through 300 fps. It was soon apparent that testing parachutes at very low velocities in the horizontal position resulted in unrealistic, inflated canopy shapes which cast doubts on the validity of the data. The minimum test velocity was raised to 50 fps where the inflated shape was well defined. However, one configuration, a series No. 2 parachute with 8 suspension lines of 1.8 L length, did provide acceptable data as shown in Figure 44. The sharp drag rise in the low-velocity range is evident. Similar trends were seen on other models. Tests in the low-velocity range could be conducted in a vertical wind tunnel where any changes in canopy shape will be realistic.